

A hierarchical iterative solver for the Navier-Stokes equations

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Joint works with:

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Outline

- Discretization of Navier-Stokes Equations
- Iterative Method
 - Preconditioning of the matrix associated with the velocity part
 - Schur complement approximation
- Numerical Simulations
 - Fluid flow around rigid objects
 - Symmetry breaking in a sudden axisymmetric constriction
 - High Performance Computing (HPC)
- Conclusion and Ongoing Works

Discretization of NSE

Navier Stokes Equations

$$\left\{ \begin{array}{l} \rho \frac{\partial \mathbf{u}}{\partial t} - \operatorname{div}(2\eta \varepsilon(\mathbf{u})) + \rho(\mathbf{u} \cdot \nabla \mathbf{u}) + \nabla p = \mathbf{f} \\ \operatorname{div}(\mathbf{u}) = 0 \\ \mathbf{u}(x, 0) = \mathbf{u}_0(x) \quad \text{such that} \quad \operatorname{div}(\mathbf{u}_0) = 0 \end{array} \right.$$

with boundary conditions

$\varepsilon(\mathbf{u}) = (\nabla \mathbf{u} + (\nabla \mathbf{u})^t)/2$ is the rate of strain tensor.

Discretization of NSE

Time Discretization

$$t_n = t_0 + n\Delta t$$

We use the backward second-order accurate implicit scheme :

$$\frac{\partial \mathbf{u}}{\partial t} \approx \frac{3\mathbf{u}^n - 4\mathbf{u}^{n-1} + \mathbf{u}^{n-2}}{2\Delta t}$$

Thus, at each time step, we obtain the **non-linear** equation:

$$\alpha \mathbf{u}^n - \operatorname{div}(2\eta \varepsilon(\mathbf{u}^n)) + \rho(\mathbf{u}^n \cdot \nabla \mathbf{u}^n) + \nabla p^n = \mathbf{f} + \alpha \mathbf{u}^*$$

$$\operatorname{div}(\mathbf{u}^n) = 0$$

where the right hand side depends only on \mathbf{u}^{n-1} and \mathbf{u}^{n-2}

Discretization of NSE

Treatment of the non-linear term

At time step,

$$\alpha \mathbf{u}^n - \operatorname{div}(2\eta \varepsilon(\mathbf{u}^n)) + \rho(\mathbf{u}^n \cdot \nabla \mathbf{u}^n) + \nabla p^n = \mathbf{f} + \alpha \mathbf{u}^*$$

This non-linearity can be treated by several approach :

- Linearized method $\mathbf{u}^n \cdot \nabla \mathbf{u}^n \approx \mathbf{u}_e \cdot \nabla \mathbf{u}^n$ (for instance: $\mathbf{u}_e = 2\mathbf{u}^{n-1} - \mathbf{u}^{n-2}$)
- Fixed point method (Picard method):
 - Given (\mathbf{u}_j^n, p_j^n) a solution at iteration j,
 - Obtain $(\mathbf{u}_{j+1}^n, p_{j+1}^n)$ a solution of

$$\alpha \mathbf{u}_{j+1}^n - \operatorname{div}(2\eta \varepsilon(\mathbf{u}_{j+1}^n)) + \rho(\mathbf{u}_j^n \cdot \nabla \mathbf{u}_{j+1}^n) + \nabla p_{j+1}^n = \mathbf{f} + \alpha \mathbf{u}^*$$

Discretization of NSE

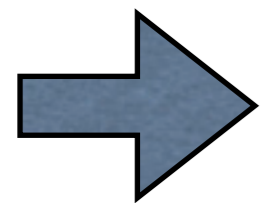
Treatment of the non-linear term

- Newton method:

at “Newton” iteration j, the approximation of the convection term:

$$\mathbf{u}_{j+1}^n \cdot \nabla \mathbf{u}_{j+1}^n \approx \delta \mathbf{u} \cdot \nabla \mathbf{u}_j^n + \mathbf{u}_j^n \cdot \nabla (\delta \mathbf{u}) + \mathbf{u}_j^n \cdot \nabla \mathbf{u}_j^n$$

where $\mathbf{u}_{j+1}^n = \mathbf{u}_j^n + \delta \mathbf{u}.$



$$\alpha \mathbf{u}_{j+1}^n - \operatorname{div}(2\eta \varepsilon(\mathbf{u}_{j+1}^n)) + \rho(\mathbf{u}_j^n \cdot \nabla \delta \mathbf{u} + \delta \mathbf{u} \cdot \nabla \mathbf{u}_j^n) +$$
$$\nabla p_{j+1}^n = \mathbf{f} + \alpha \mathbf{u}^* - \rho(\mathbf{u}_j^n \cdot \nabla) \mathbf{u}_j^n$$

Discretization of NSE

Space Discretization

The discrete subspaces \mathbf{V}_h and Q_h are chosen as follows:

$$\mathbf{V}_h = \{ \mathbf{v}_h \in C^0(\Omega)^d, \mathbf{v}_h|_T \in (P_2)^d \quad \forall T \in \mathcal{T}_h \}$$

$$Q_h = \{ q_h \in C^0(\Omega), q_h|_T \in (P_1) \quad \forall T \in \mathcal{T}_h \}$$

Then, the velocity and pressure can be decomposed as follows:

$$\mathbf{u}_h = \sum_{i=1}^n u_i \boldsymbol{\phi}_i \quad \text{and} \quad p_h = \sum_{j=1}^m p_j \psi_j$$

Furthermore, the hierarchical basis allows the follows decomposition

$$V_h = V_1 \oplus V_q \quad (u = u_1 + u_q)$$

Where: $\left\{ \begin{array}{l} V_1 \text{ is the subspace of continuous piecewise linear polynomials, and} \\ V_q \text{ is the complementary subspace of continuous piecewise quadratic} \end{array} \right.$

Discretization of NSE

Algebraic System

Finally, the discrete formulation can be rewritten as:

$$\mathcal{A} \begin{bmatrix} u \\ p \end{bmatrix} = \begin{bmatrix} F & B^t \\ B & 0 \end{bmatrix} \begin{bmatrix} u \\ p \end{bmatrix} = \begin{bmatrix} f \\ 0 \end{bmatrix}$$

Where $F = \alpha M + \eta D + \rho(C + N)$ such that

- *Mass matrix* $M_{ij} = \int_{\Omega} \phi_i \cdot \phi_j \, dx$
- *Diffusion matrix* $D_{ij} = \int_{\Omega} \varepsilon(\phi_i) : \varepsilon(\phi_j) \, dx$
- *Convection matrix* $C_{ij} = \int_{\Omega} (\mathbf{w} \cdot \nabla \phi_i) \cdot \phi_j \, dx$ $N_{ij} = \int_{\Omega} \underbrace{(\phi_i \cdot \nabla \mathbf{w}) \cdot \phi_j}_{\text{From Newton Method}} \, dx$
- *Divergence matrix* $B_{ij} = - \int_{\Omega} \psi_j \operatorname{div}(\phi_i) \, dx$

Resolution Method

Direct Method

To solve the system:

$$\begin{bmatrix} F & B^t \\ B & 0 \end{bmatrix} \begin{bmatrix} u \\ p \end{bmatrix} = \begin{bmatrix} f \\ 0 \end{bmatrix}$$

we can use the direct method

- The matrix is singular, then we introduce a penalization term

$$\begin{bmatrix} F & B^t \\ B & \epsilon M \end{bmatrix} \begin{bmatrix} u \\ p \end{bmatrix} = \begin{bmatrix} f \\ 0 \end{bmatrix}$$

- But,

Resolution Method

Iterative Method

Our approach consists:

- Solve simultaneously the velocity and the pressure
- GCR or GMRES (Krylov method)
- Preconditioner:

$$\begin{bmatrix} F & B^t \\ B & 0 \end{bmatrix} = \begin{bmatrix} I & 0 \\ B F^{-1} & I \end{bmatrix} \begin{bmatrix} F & 0 \\ 0 & -S \end{bmatrix} \begin{bmatrix} I & F^{-1} B^t \\ 0 & I \end{bmatrix}$$

where $S = B F^{-1} B^t$ stands for the Schur complement

This factorization cannot be used as preconditioner because
F is large-scale matrix and S is dense matrix.

Iterative Method

Iterative Method

- Then, we introduce an approximation: \tilde{F} and \tilde{S}
- And, we choose the follow preconditioner:

$$\mathcal{P}_R = \begin{bmatrix} \tilde{F} & B^t \\ 0 & -\tilde{S} \end{bmatrix}$$

- The action of this preconditioner on a residual vector can be rewritten as:

$$\begin{cases} \delta p = -\tilde{S}^{-1} r_p \\ \delta u = \tilde{F}^{-1} (r_u - B^t \delta p) \end{cases}$$

We have to solve 2 systems

Iterative Method

Preconditioning the matrix F

$$\delta u = \tilde{F}^{-1}(r_u - B^t \delta p) \quad \Leftrightarrow \quad F \delta u = \tilde{r}_u$$

We use the hierarchical basis for the quadratic FE discretization

Then, we can decompose the velocity into the linear part and a quadratic correction:

$$u = u_l + u_q$$

Consequently, the matrix F can be rewritten as:
$$F = \begin{bmatrix} F_{ll} & F_{lq} \\ F_{ql} & F_{qq} \end{bmatrix}$$

Finally, we use the following Algorithm proposed by El Maliki and Fortin

1. Solve by few iterations of SOR: $F \delta = r$ where $\delta = (\delta_l, \delta_q)^t$ and $r = (r_l, r_q)^t$.
2. Compute the residual: $d_l = r_l - F_{ll} \delta_l - F_{lq} \delta_q$.
3. Solve by a direct or few iterations of an iterative method: $F_{ll} \delta_l^* = d_l$.
4. Update the correction: $\delta = (\delta_l + \delta_l^*, \delta_q)^t$.

Iterative Method

Schur complement approximation

Thanks to the discrete inf-sup condition proved by Brezzi-Fortin:

$$\xi^2 \leq \frac{p^t (B D^{-1} B^t) p}{p^t M_p p} \leq \chi^2$$

we can remark that the matrix $B D^{-1} B^t$ is spectrally equivalent to the mass matrix M_p

- *Stokes Case:* We can choose this approximation

$$S \simeq \tilde{S} = \frac{1}{\eta} M_p \simeq \frac{1}{\eta} \text{diag}(M_p)$$

- *Navier Stokes Case:* Turek proposes the additive preconditioner:

$$S^{-1} \simeq M_p^{-1} (\alpha M_p + \eta D_p + \rho C_p) D_p^{-1}$$

Numerical Simulations with MEF++ Code

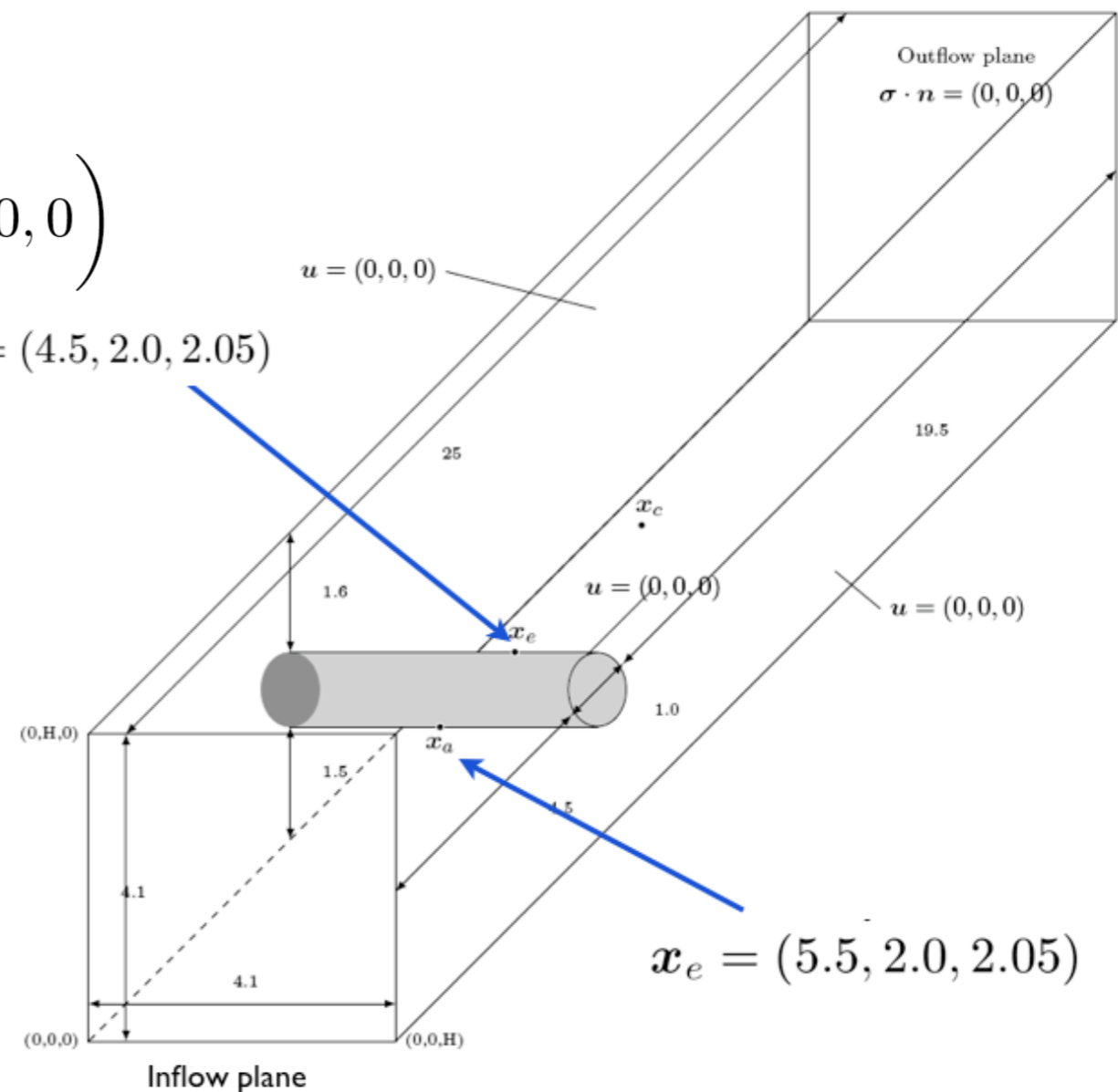
Fluid flow around rigid objects

- Configuration and boundary conditions for flow around cylinder*

The inflow condition is:

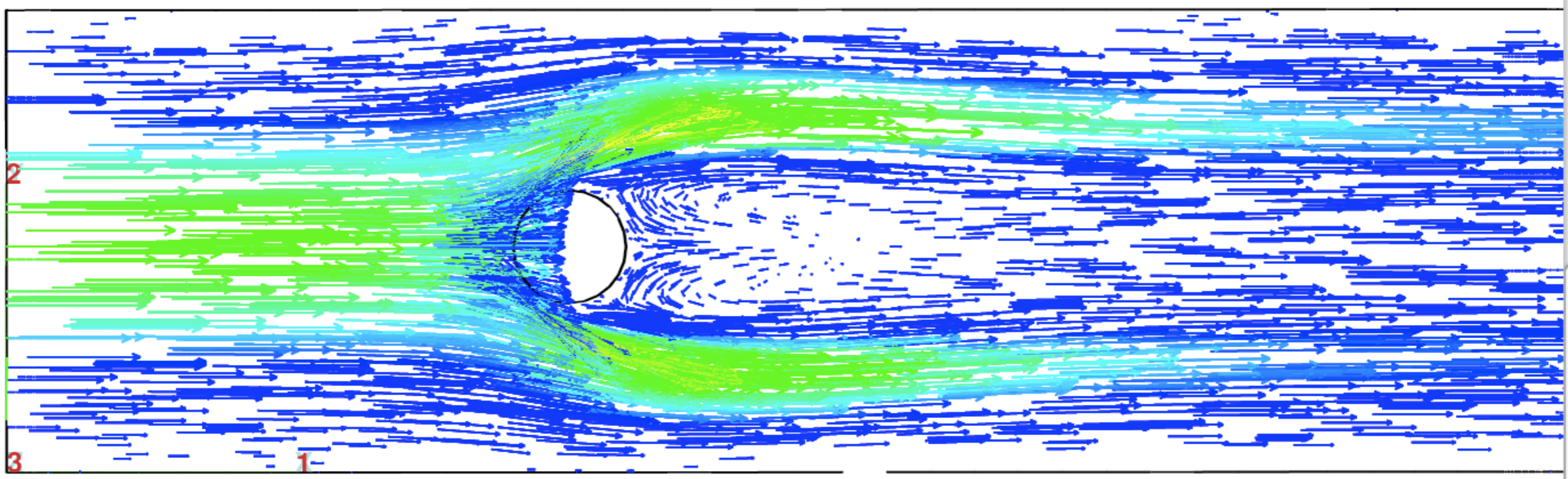
$$\mathbf{u}(0, x_2, x_3) = \left(\frac{72}{H^4} x_2 x_3 (H - x_2)(H - x_3), 0, 0 \right)$$

$$Re = 20.$$



Numerical Simulations with MEF++ Code

Fluid flow around rigid objects



- *Comparison with Turek, Schafer,...*

	<i>Our Results</i>	<i>Turek Teams Results</i>
Δp	0.1694	0.1693
C_D	6.1430	6.0928

Numerical Simulations with MEF++ Code

Fluid flow in a sudden axisymmetric constriction

- *Geometry*



- *Physical experience*

- **J. Vetel and A. Garon**

Symmetry breaking

at Reynolds Number = 255.

- *Stability Method*

- **S. J. Sherwin et al**

Symmetry breaking

at Reynolds Number = 721.

Numerical Simulations with MEF++ Code

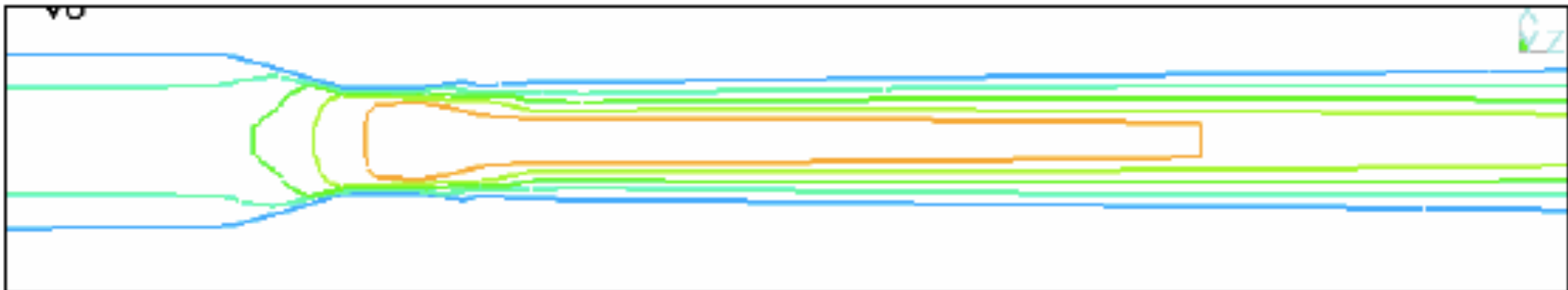
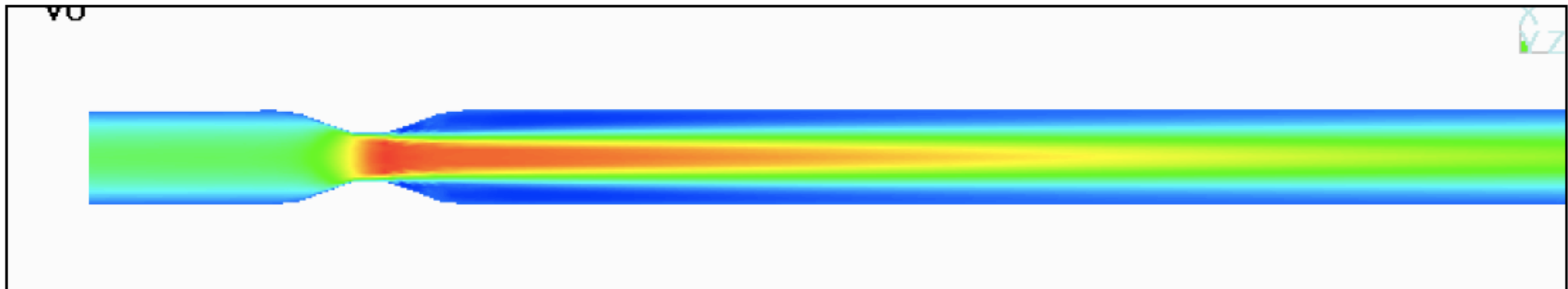
Fluid flow in a sudden axisymmetric constriction

- *Our Strategy!* (*Reynolds number = 300*)
 - We consider a symmetric Mesh
 1. By a Reynolds Continuation Procedure, we obtain a symmetric solution
 2. Build a non-symmetric solution by imposing some boundary conditions
 3. Redo the simulation by setting this non-symmetric as initial data \mathbf{u}_0

Numerical Simulations with MEF++ Code

Fluid flow in a sudden axisymmetric constriction

1. Symmetric Solution at Reynolds=300

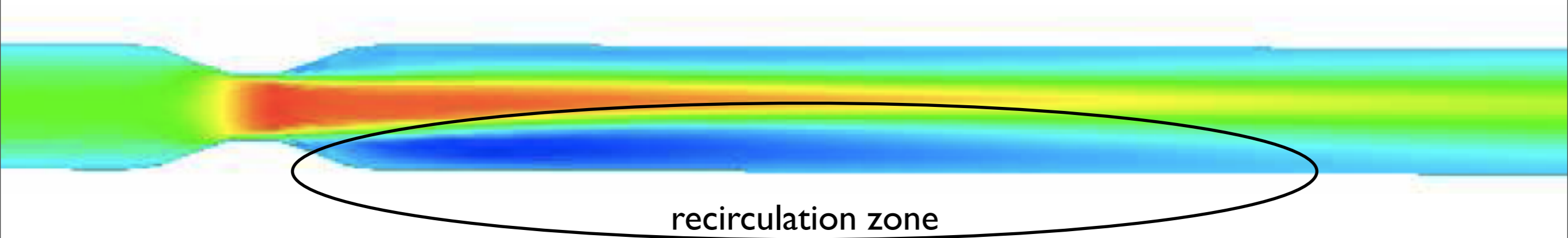
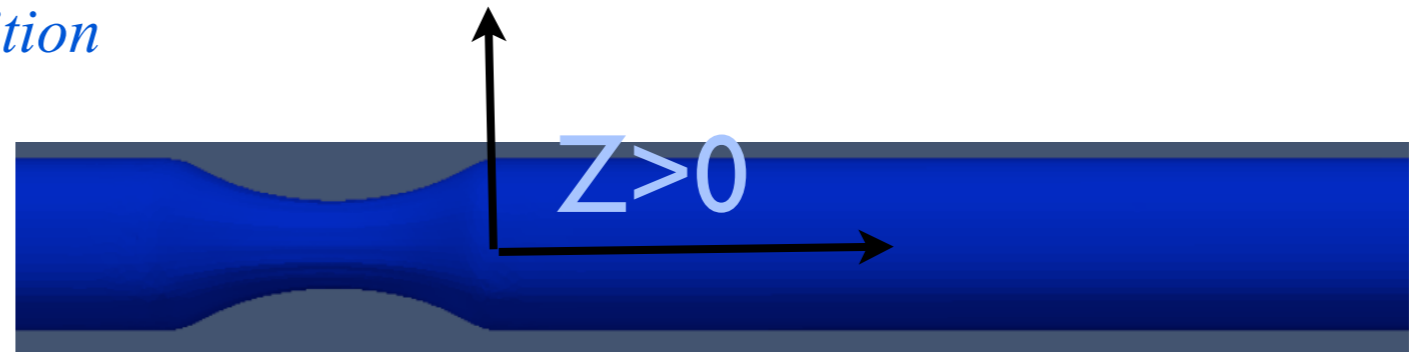


Numerical Simulations with MEF++ Code

Fluid flow in sudden axisymmetric constriction

2. Perturbation of the Initial Condition

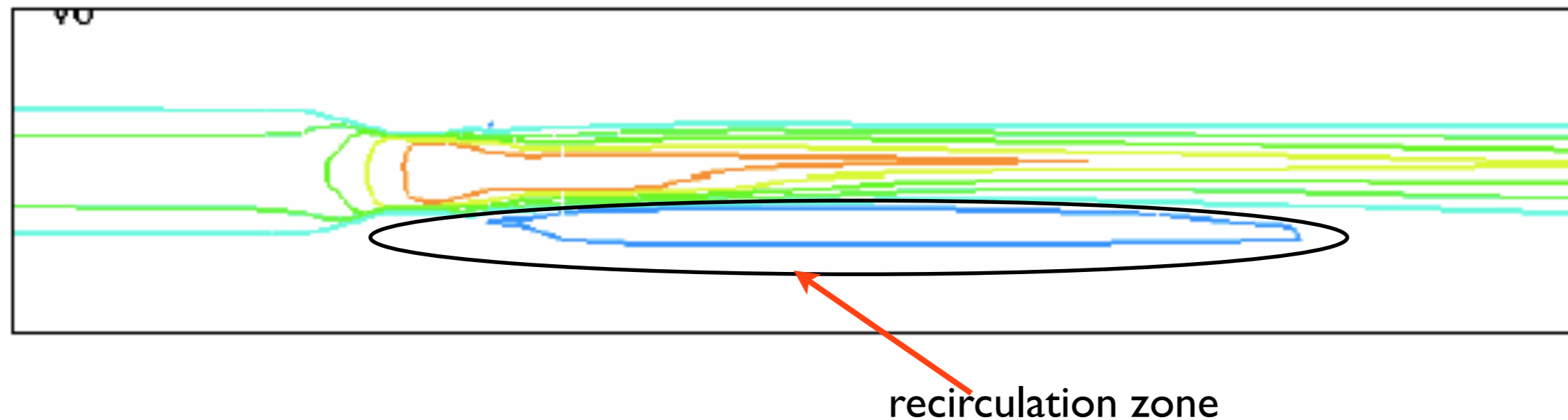
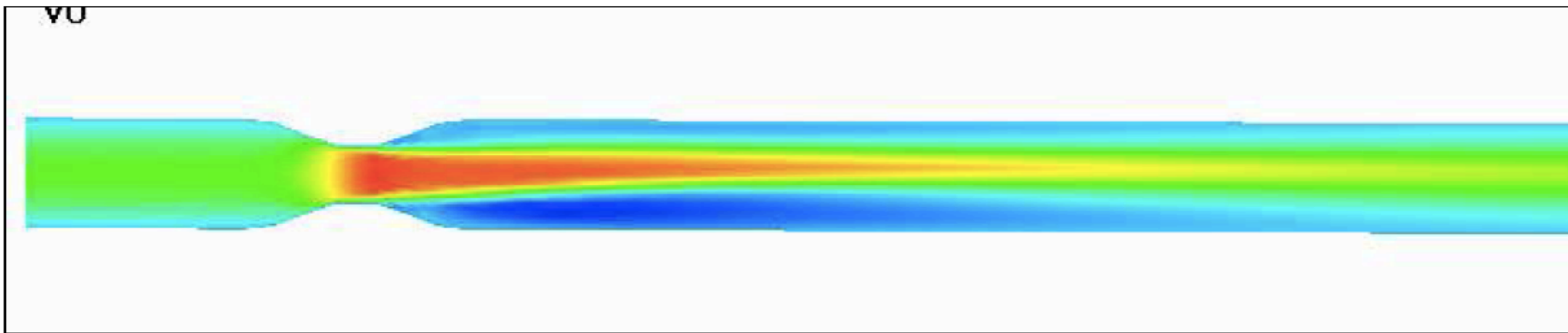
- Boundary condition:
 $z > 0, y > 0: u(x, y, z) = (0, 0, \text{free})$ and
if $z \leq 0$ or $y \leq 0: u(x, y, z) = (0, 0, 0)$.



Numerical Simulations with MEF++ Code

Fluid flow in sudden axisymmetric constriction

3. Existence of the asymmetric solution at Reynolds = 300



High Performance Computing

MEF++ //

colosse.clumeq.ca: 7900 Cores

High Performance Computing

MEF++ //

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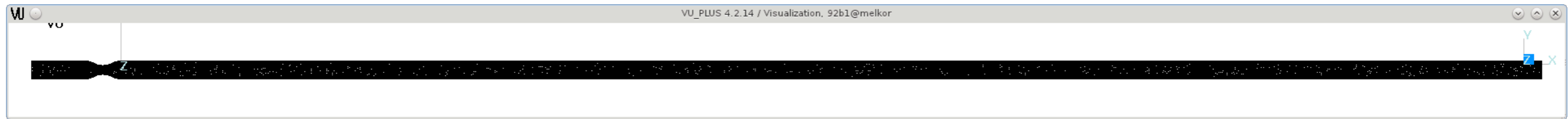
- *4.5M DOF*

High Performance Computing

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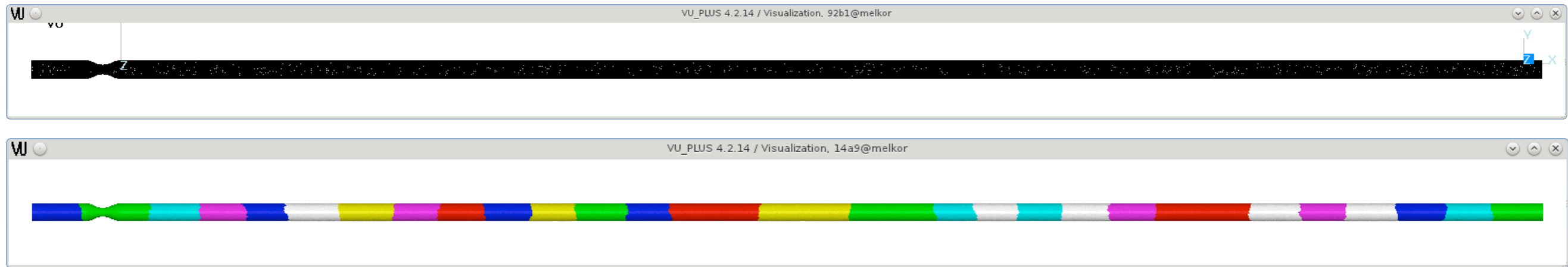


High Performance Computing

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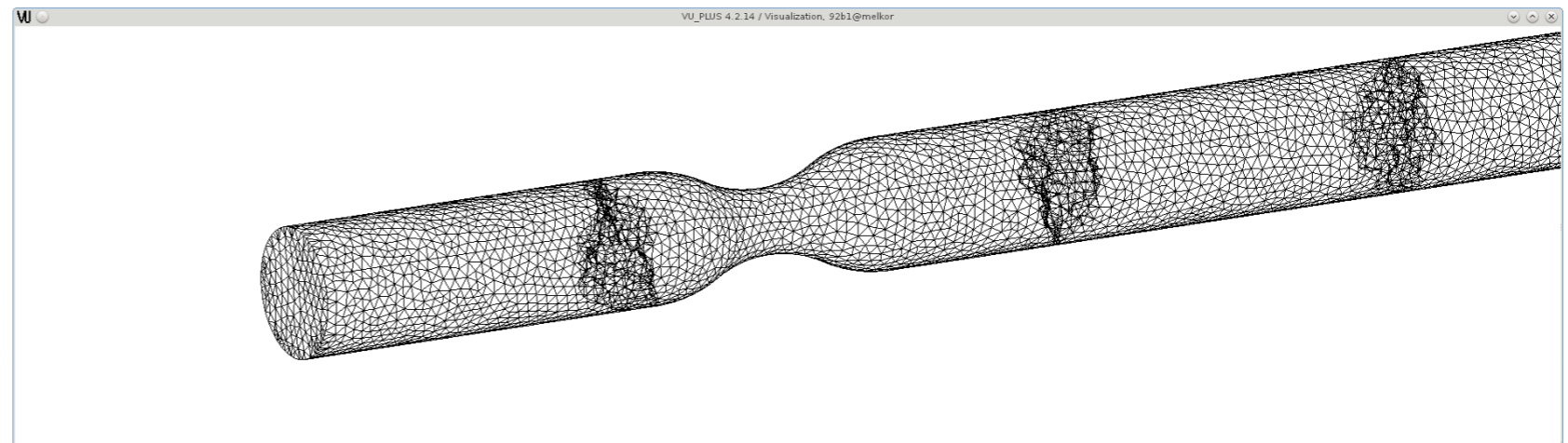
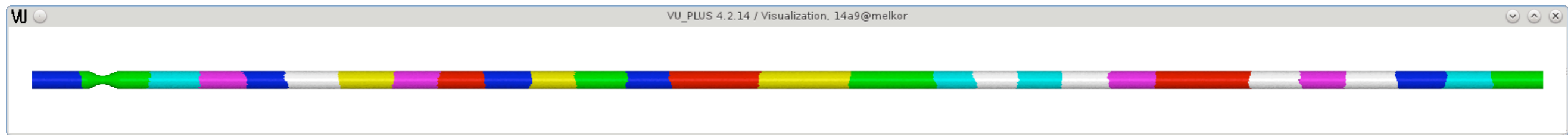
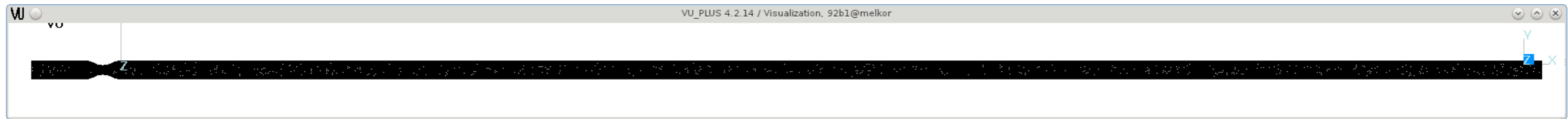


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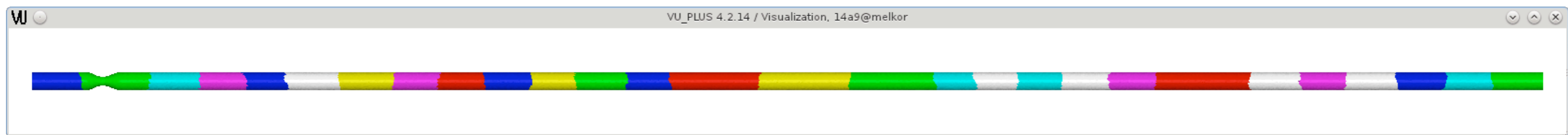
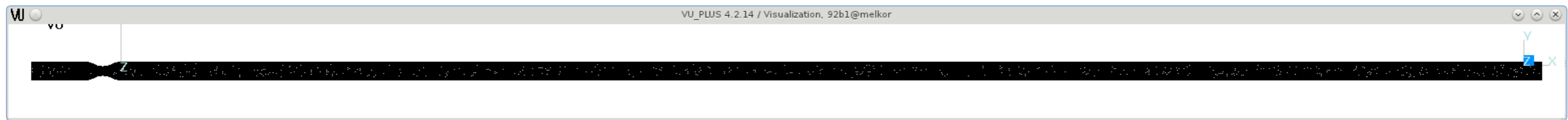


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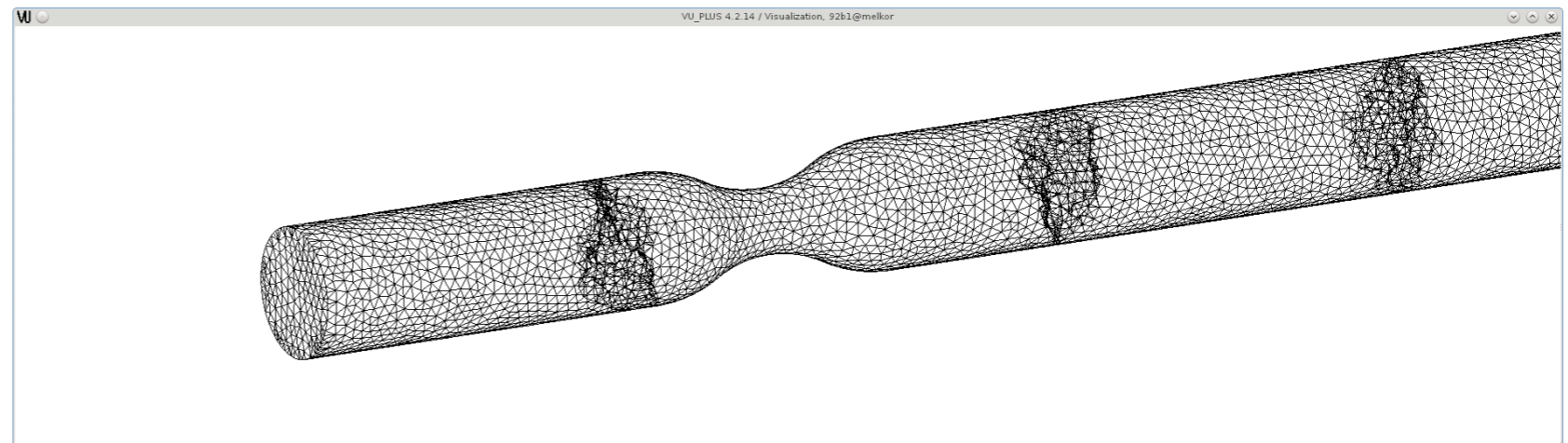
MEF++ //

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- *4.5M DOF*



For 3 time steps, 10 Iterations of Newton

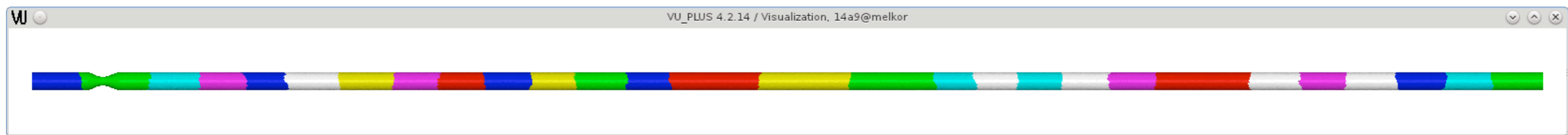
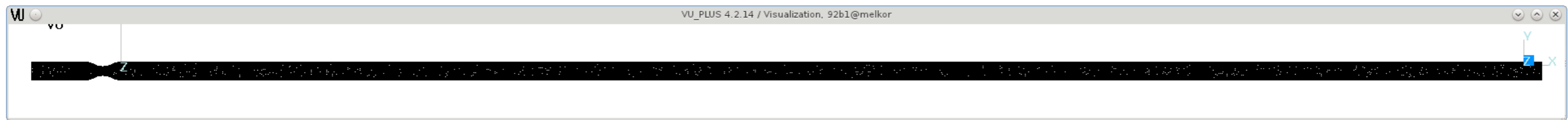


High Performance Computing

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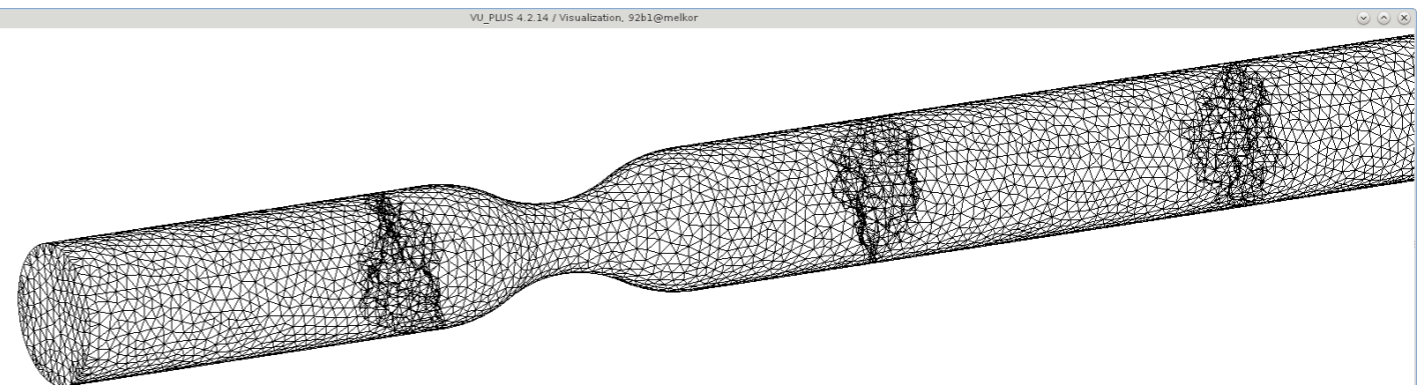
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- **4.5M DOF**



For 3 time steps, 10 Iterations of Newton

DOF	Nb_proc	Time(s)
1 403 216	2	2002
2 790 733	4	2394
4 929 523	8	2682
10 281 710	14	3539
47 083 332	64	5939



Conclusion and Ongoing Works

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 2. MEF++ Code is Parallel
 3. We can simulate big problems

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1. Verify experimental results of Vetel and Garon (and Sherwin)
2. Improvement (optimization) of the solver
3. Fluid-Structure Interaction

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Thank you!